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FAILURE MODES EFFECTS ANALYSIS (FMEA) - NON-CIL HARDWARE NUMBER:05-6PB-22107M -X

SUBSYSTEM NAME: EPD&C - C&T; UHF SPACE COMMUNICATION

REVISION: 0

11/14/95

PART DATA

PARTNAME VENDOR NAME

PART NUMBER VENDOR NUMBER

LRU : PANEL R15

VO70-730325

SRU

: CIRCUIT BREAKER

MC454-0026-2075

EXTENDED DESCRIPTION OF PART UNDER ANALYSIS:

CIRCUIT BREAKER, UHF - ATC, CB13

REFERENCE DESIGNATORS: 32V73A15CB13

QUANTITY OF LIKE ITEMS:

ONE

FUNCTION:

PROVIDES +28 VDC (MAIN A) POWER TO UHF - ATC TRANSCEIVER. MAIN A POWERS ALL UHF - ATC FUNCTIONS INCLUDING UHF POWER AMP.

REFERENCE DOCUMENTS: VS70-740119

FAILURE MODES EFFECTS ANALYSIS FMEA - NON-CIL FAILURE MODE

NUMBER: 05-6PB-22107M-01

REVISION#: 0

11/14/95

SUBSYSTEM NAME: EPD&C - C&T: UHF SPACE COMMUNICATION

LRU: PANEL R15

CRITICALITY OF THIS

ITEM NAME: CIRCUIT BREAKER

FAILURE MODE: 1R3

FUNCTIONAL CRITICALITY!

REQUIRED FAULT TOLERANCE/ACHIEVED FAULT TOLERANCE:1R/2/2

FAILURE MODE:

FAILS OPEN, FAILS TO CONDUCT, FAILS TO CLOSE

MISSION PHASE:

PL PRE-LAUNCH

LO LIFT-OFF

DO DE-ORBIT

LS LANDING/SAFING

VEHICLE/PAYLOAD/KIT EFFECTIVITY:

102 COLUMBIA

103 DISCOVERY

104 ATLANTIS

105 ENDEAVOUR

AFTER SPACE COMM MODIFICATION

CAUSE:

VIBRATION, TEMPERATURE, MECHANICAL SHOCK, CONTAMINATION, MISHANDLING. PIECE-PART STRUCTURAL FAILURE.

CRITICALITY 1/1 DURING INTACT ABORT ONLY? NO

CRITICALITY 1R2 DURING INTACT ABORT ONLY (AVIONICS ONLY)? NO

REDUNDANCY SCREEN

A) PASS

B) PASS

C) PASS

PASS/FAIL RATIONALE:

A)

B)

C)

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FAILURE MODES EFFECTS ANALYSIS (FMEA) -- NON-CIL FAILURE MODE NUMBER: 05-6P8-22107M-01

CORRECTING ACTION: MANUAL
CORRECTING ACTION DESCRIPTION: LOW POWER RECEPTION MAY BE POSSIBLE BY VEHICLE ORIENTATION.
- FAILURE EFFECTS -
(A) SUBSYSTEM: UNABLE TO TRANSMIT HIGH POWER SIMPLEX
(B) INTERFACING SUBSYSTEM(S): NO EFFECT FIRST FAILURE
(C) MISSION: AIR/GROUND: RANGE TO GROUND STATION LIMITED.
(D) CREW, VEHICLE, AND ELEMENT(S): NO EFFECT FIRST FAILURE
E) FUNCTIONAL CRITICALITY EFFECTS: AFTER 3 FAILURES (CIRCUIT BREAKER CB13 AND 2 S-BAND) POSSIBLE LOSS OF CREW/VEHICLE DUE TO LOSS OF STATE VECTOR UPDATE.
- TIME FRAME -
TIME FROM FAILURE TO CRITICAL EFFECT: MINUTES
- APPROVALS -
- AFFROVALS -
RODUCT ASSURANCE ENGR : VAN D. NGUYEN UMALINEN 8-20.98 ESIGN ENGINEERING : G. J. SCHWARTZ